Principles and Theory of Radar Interferometry

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UNAVCO Short Course
Outline of Tutorial

I. Quick Review of Radar Imaging Fundamentals
   A. Basic Principles of SAR
   B. Range and Azimuth Compression
   C. ScanSAR vs Strip Mode Processing

II. Geometric Aspects of Interferometric Phase Measurement
   A. Interferometric Phase for Topographic Mapping
   B. Interferometric Phase for Deformation Mapping
   C. Sensitivity of Topographic Phase Measurements
   D. Sensitivity of Deformation Phase Measurements
   E. Comparison of Various Interferometric Mapping Schemes

III. Interferometric Correlation
   A. Random and Deterministic Portions of the Interferometric Phase
   B. SNR and Interferometric Correlation
   C. Geometric Decorrelation and Range Spectral Shift
   D. Temporal Decorrelation
   E. Volumetric Decorrelation
   F. Other Error Sources

IV. Interferometric Processing
   A. Processing Flow
   B. Interferogram Formation
   C. Image Co-registration
   D. Baseline Determination
   E. Phase Unwrapping
   F. Height Determination
   G. Deformation Signal Extraction
The Radar Concept

- Much like sound waves, radar waves carry information that echoes from distant objects.
- The time delay of the echo measures the distance to the object.
- The changes of the message in the echo determines the object characteristics.

Transmitter/Receiver

At light speed, \( c \)

Object scatters energy back to radar
Radar on a Moving Platform

- Pulses are transmitted from the radar platform as it moves along its flight path.
- Each pulse has finite extent in time, illuminating a narrow strip of ground as it sweeps through the antenna beam.
- Some of the energy from the ground is scattered back to the radar instrument.

**Antenna Beamwidth**

\[ \theta_{3dB} = \frac{\lambda}{L} \]

**Footprint Size on Ground**

\[ \Delta X_{az} \approx \rho \theta_{3dB_{az}} = \frac{\rho \lambda}{L_{az}} \]

\[ \Delta X_{\rho} \approx \rho \theta_{3dB_{\rho}} = \frac{\rho \lambda}{L_{\rho}} \]
Matched Filtering of Received Echo

Point target

- Transmitted pulses are usually coded waveforms with significant bandwidth
- Matched filtering allows recovery of fine resolution features with a low peak-power pulse train
Imaging Radar

Small antenna - big beam

Big antenna - small beam

Synthesized long antenna

CROSS-TRACK RESOLUTION ACHIEVED BY SHORT PULSE LENGTHS (HIGH BANDWIDTH)

ALONG-TRACK RESOLUTION ACHIEVED BY COHERENTLY COMBINING ECHOES FROM MULTIPLE PULSES ALONG-TRACK (SYNTHESIZE A LONG ANTENNA)

- RESOLUTION $\propto$ ANTENNA LENGTH
- INDEPENDENT OF RANGE/FREQUENCY
SAR Imaging Concept

Cross-Track resolution achieved by short pulse lengths (High Bandwidth)

Along-Track resolution achieved by coherently combining echoes from multiple pulse along-track (Synthesize a long antenna)
- Resolution proportional to antenna length
- Independent of Range/Frequency
SAR Imaging Concept

Cross-Track resolution achieved by short pulse lengths (High Bandwidth)

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SAR Imaging Concept

Cross-Track resolution achieved by short pulse lengths (High Bandwidth)

Real Along Track Beam
Synthesized Along Track Beam

Along-Track resolution achieved by coherently combining echoes from multiple pulse along-track (Synthesize a long antenna)
- Resolution proportional to antenna length
- Independent of Range/Frequency
Azimuth Resolution from Aperture Synthesis

- The synthetic aperture $X_{\text{ill}}$ lengthens as $R_0$ increases…
- …which decreases the azimuth synthetic aperture’s angular beamwidth $(\lambda/X_{\text{ill}})$ in proportion…
- …but the spatial resolution on the ground $(\lambda R_0/X_{\text{ill}} = L/2)$ is constant
- Aperture synthesis processing is very similar to matched...
In ScanSAR the look angle for a burst of pulses changes for each dwell period.

burst of pulses in alternating dwells...

...compared to a continuous strip of pulses at fixed look angle.
Illustration of Burst Combination

Images formed from each burst overlap in the ground plane.

Images are incoherently added (in power) to recover looks in processing.
Strip versus ScanSAR images (no radiometric calibration)

Standard Strip Mode Amplitude (4 looks; then 4x4 more)

Burst Processed Amplitude (1/4 aperture; 2 looks; then 4x4 more)

Note Amplitude Scalloping
What Radar Can Tell Us

• Transmitted radar signals have known characteristics
  – Amplitude
  – Polarization
  – Phase and Time Reference
  – Wavelength, or Frequency

• A distant object that scatters the radar signal back toward the receiver alters the amplitude, polarization, and phase, differently for different wavelengths

• Comparison of the received signal characteristics to the transmitted signal allows us to understand the properties of the object.

• This is the principle of active remote sensing.
Radar Imaging Properties

- Radar images are distorted relative to a planimetric view. Slopes facing toward or away from the radar appear foreshortened. Steep slopes are collapsed into a single range cell called layover and areas occulted by other areas are said to be shadowed.

- Radar is primarily sensitive to the structure of objects being imaged whereas optical images are primarily sensitive to chemistry.

- The scale of objects relative to the radar wavelength determines how smooth an object appears to the radar and how bright or dark it is in the imagery.
Wavelengths - A Measure of Surface Scale Sizes

Light interacts most strongly with objects on the size of the wavelength.

**Forest:** Leaves reflect X-band wavelengths but not L-band.

**Dry soils:** Surface looks rough to X-band but not L-band.

**Ice:** Surface and layering look rough to X-band but not L-band.
Radar Images in Different Bands

X-band

L-band

P-band
Polarization - A Measure of Surface Orientations and Properties

Wave Polarization

Vertical polarization passes through horizontally arranged absorbers.

Horizontal polarization does not pass through horizontally arranged absorbers.

Mostly horizontal polarization is reflected from a flat surface.

Color figures from www.colorado.edu/physics/2000
Polarimetry from SIR-C/X-SAR

In red areas, streets are oriented along the shuttle track.

The geometry of the surface is captured in the polarization.
Phase - A Measure of the Range and Surface Complexity

The phase of the radar signal is the number of cycles of oscillation that the wave executes between the radar and the surface and back again.

The total phase is two-way range measured in wave cycles + random component from the surface.

Collection of random path lengths jumbles the phase of the echo.

Only interferometry can sort it out!
What is Radar Interferometry?

- Radar interferometry can be broadly defined by use of phase measurements to precisely measure the relative distance to an object when imaged by synthetic aperture radar from two or more observations separated either in time or space.
  - Interferometric phase is simply another means of measuring distance.

\[ \phi = \frac{2\pi \Delta \rho}{\lambda} \]

- Phase measurements in interferometric systems can be made with degree level accuracy, and with typical radar wavelengths in 3-80 cm range this corresponds to relative range measurements having millimeter accuracy.
Interference Concept

- Interference occurs when the phase of two different waves is not aligned. The observed intensity, $I$, is the time average of the sum of the wave fields.

\[ E(z,t) = E_0 \cos(2\pi(f t - z / \lambda)) \]

Phase aligned waves add constructively.

Phase opposed waves add destructively.

Intermediate phase alignment.

Fig. 7.1. Interference of two beams of equal intensity; variation of intensity with phase difference.
Young’s Interferometer

- In Young’s experiment, a point source illuminates two separated vertical slits in an opaque screen. The slits are very narrow and act as line sources. For this case, the pattern of intensity variations on the observing screen is bright/dark banding.

Fig. 7.2. Young’s experiment
(Born and Wolf, 1980)
Young’s Interferometer Geometry

- The brightness variations can be understood in terms of the relative phase of the interfering waves at the observing screen.
- The spacing of fringes is set by the slit separation.
- Phase = $2 \pi \times d / a \lambda$
- $x_{\text{max}} = m a \lambda / d$, $m = 0, 1, 2, ...$

Fig. 7.3. Illustrating interference with two point sources.
(Born and Wolf, 1980)
Radar Interferometry

- Radar Interferometry is a simple extension of the Young’s interferometry concept.
- Radar has a coherent source much like a laser.
- The two radar (SAR) antennas act as coherent point sources.
- Because the wavelengths are so long, the signal can easily be digitized and processed coherently, measuring the phase information directly.
- When imaging a surface, the phase fronts from the two sources interfere.
- The surface topography slices the interference pattern.
- The measured phase differences record the topographic information.
Radar Interferometry Example

Standard Radar Image

Interference fringes follow the topography

One cycle of color represents 1/2 wavelength of path difference
Types of Radar Interferometry

- Two main classes of interferometric radars are separated based on the geometric configuration of the baseline vector, i.e. the vector separating the antenna locations in the interferometric pair. These are:
  - Cross-Track interferometers used for topographic and surface deformation measurements whereby the antennas are nominally separated in the cross-track direction.
  - Along-Track interferometers used to measure radial velocity whereby the antennas are separated in the along-track direction.

Cross-Track Interferometer

Along-Track Interferometer

- Dual antenna single pass interferometers
- Single antenna repeat pass interferometers
  
  \[ \Rightarrow \text{Topography and Deformation} \]

- Dual antenna single pass interferometer
- Along-track separation
  
  \[ \Rightarrow \text{Radial velocity} \]
Interferometry Applications

• Mapping/Cartography
  – Radar Interferometry from airborne platforms is routinely used to produce topographic maps as digital elevation models (DEMs).
    • 2-5 meter circular position accuracy
    • 5-10 m post spacing and resolution
    • 10 km by 80 km DEMs produced in 1 hr on mini-supercomputer
  – Radar imagery is automatically geocoded, becoming easily combined with other (multispectral) data sets.
  – Applications of topography enabled by interferometric rapid mapping
    • Land use management, classification, hazard assessment, intelligence, urban planning, short and long time scale geology, hydrology

• Deformation Mapping and Change Detection
  – Repeat Pass Radar Interferometry from spaceborne platforms is routinely used to produce topographic change maps as digital displacement models (DDMs).
    • 0.3-1 centimeter relative displacement accuracy
    • 10-100 m post spacing and resolution
    • 100 km by 100 km DDMs produced rapidly once data is available
  – Applications include
    • Earthquake and volcano monitoring and modeling, landslides and subsidence
    • Glacier and ice sheet dynamics
    • Deforestation, change detection, disaster monitoring
Interferometry for Topography

Measured phase difference:

\[ \phi = -\frac{2\pi}{\lambda} \delta \rho \]

Triangulation:

\[ \sin(\theta - \alpha) = \frac{(\rho + \delta \rho)^2 - \rho^2 - B^2}{2 \rho B} \]

\[ z = h - \rho \cos \theta \]

Critical Interferometer Knowledge:
- Baseline, \((B, \alpha)\), to mm’s
- System phase differences, to deg’s
Height Reconstruction

• Interferometric height reconstruction is the determination of a target’s position vector from known platform ephemeris information, baseline information, and the interferometric phase.

\[ \vec{P} = \text{platform position vector} \]
\[ \rho = \text{range to target} \]
\[ \hat{l} = \text{unit vector pointing from platform to target} \]
\[ \vec{T} = \text{target location vector} \]

**BASIC EQUATION**

\[ \vec{T} = \vec{P} + \rho \hat{l} \]

• Interferometry provides a means of determining \( \hat{l} \).
Interferometric Geometry

\[ \vec{b} = \vec{P}_2 - \vec{P}_1 = \vec{l}_1 - \vec{l}_2 \]
\[ \vec{l}_2 = \vec{l}_1 - \vec{b} \]
\[ b = |\vec{b}| = \langle \vec{b}, \vec{b} \rangle^{\frac{1}{2}} \]
\[ \hat{l}_1 = \vec{l}_1 = \frac{\vec{l}_1}{|\vec{l}_1|} = \frac{\vec{l}_1}{\rho_1} \]

\[ \phi = \frac{2\pi p}{\lambda} (\rho_2 - \rho_1) = \frac{2\pi p}{\lambda} (|\vec{l}_2| - |\vec{l}_1|) \]
\[ = \frac{2\pi p}{\lambda} \left( \langle \vec{l}_2, \vec{l}_2 \rangle^{\frac{1}{2}} - \rho_1 \right) \]
\[ = \frac{2\pi p}{\lambda} \left( \langle \vec{l}_1 - \vec{b}, \vec{l}_1 - \vec{b} \rangle^{\frac{1}{2}} - \rho_1 \right) \]
\[ = \frac{2\pi p}{\lambda} \left( \rho_1^2 - 2\langle \vec{l}_1, \vec{b} \rangle + b^2 \right)^{\frac{1}{2}} - \rho_1 \]
\[ = \frac{2\pi p}{\lambda} \rho_1 \left( 1 - \frac{2\langle \vec{l}_1, \vec{b} \rangle}{\rho_1} + \left( \frac{b}{\rho_1} \right)^2 \right)^{\frac{1}{2}} - 1 \]

- \( p \) equals 1 or 2 depending on system
2-D Height Reconstruction - Flat Earth

Before considering the general 3-D height reconstruction it is instructive to first solve the two dimensional problem.

Assume that $b \ll \rho$ and let $\hat{l}_1 = \frac{\hat{l}}{|\hat{l}|} = \frac{\hat{l}}{\rho}$. Taking a first order Taylor’s expansion of

$$\phi = \frac{2\pi p}{\lambda} \rho_1 \left( \left( 1 - \frac{2 \langle \hat{l}_1, \bar{b} \rangle}{\rho_1} \right) + \left( \frac{b}{\rho_1} \right)^2 \right)^{\frac{1}{2}} - 1$$

the interferometric phase can be approximated as

$$\phi \approx -\frac{2\pi p}{\lambda} \langle \hat{l}_1, \bar{b} \rangle$$

With $\bar{b} = (b \cos(\alpha), b \sin(\alpha))$ and $\hat{l} = (\sin(\theta), -\cos(\theta))$ then

$$\phi = -\frac{2\pi p}{\lambda} b \sin(\theta - \alpha)$$
Let \( \vec{P} = (y_o, h) \) be the platform position vector, then

\[
\vec{T} = \vec{P} + \rho \hat{\ell} = (y_o, h) + \rho (\sin(\theta), -\cos(\theta)) = (y_o + \rho \sin(\theta), h - \rho \cos(\theta))
\]

Solving for \( \theta \) in terms of the interferometric phase, \( \phi \), yields

\[
\theta = \sin^{-1}\left(\frac{-\lambda \phi}{2\pi pb}\right) + \alpha
\]
Phase Gradient

• For a number of applications including flight planning and unwrapping studies it is desirable to be able to compute the interferometric phase gradient for an arbitrarily sloped surface, look geometry and baseline.

• The interferometric phase is well approximated for most applications by

\[ \phi \approx -\frac{2\pi p}{\lambda} \langle \hat{b}, \hat{l} \rangle \]

where \( \vec{b} \) is the baseline vector, \( \hat{l} \) a unit vector pointing to the target and \( \lambda \) is the wavelength.

• The phase gradient is

\[ \nabla \phi = \begin{bmatrix} \frac{\partial \phi}{\partial s} \\ \frac{\partial \phi}{\partial \rho} \end{bmatrix} = -\frac{2\pi p}{\lambda} \begin{bmatrix} \langle \hat{b}, \frac{\partial \hat{l}}{\partial s} \rangle \\ \langle \hat{b}, \frac{\partial \hat{l}}{\partial \rho} \rangle \end{bmatrix} \]

where \( s \) is the along track coordinate and \( \rho \) is the range.
Phase Gradient Observations

• There in a change in phase with respect to range regardless of whether the terrain is sloped in the range direction or not. The phase rate, or fringe rate, with respect to a flat surface is called the flat surface (or spherical earth) fringe frequency.

\[
\frac{\partial \phi}{\partial \rho} = \frac{2\pi p}{\lambda} \frac{b \cos(\theta - \alpha)}{\rho \tan(\theta - \psi_c)} = \frac{2\pi p}{\lambda} \frac{b_\perp}{\rho \tan(\theta - \psi_c)}
\]

• Note that the fringe rate depends on the local slope and the perpendicular baseline length.

• The fringe rate in the azimuth or along track direction is zero unless there is an azimuth slope. It also depends on the magnitude of the local slope and the perpendicular baseline length.

\[
\frac{\partial \phi}{\partial s} = \frac{2\pi p}{\lambda} \frac{b \cos(\theta - \alpha)}{\rho \sin \theta \tan \psi_s} = \frac{2\pi p}{\lambda} \frac{b_\perp}{\rho \sin \theta \tan \psi_s}
\]
Flattened Phase

- Often when looking at interferograms or prior to unwrapping it is desirable to remove the flat earth fringes so that the resulting fringes will follow the local topography. This process is called flattening and the resulting phase is called the flattened phase.
- The flattened phase is given by

\[ \phi_{flat} = -\frac{4\pi}{\lambda} \left( \langle \hat{\ell}, \vec{b} \rangle - \langle \hat{\ell}_o, \vec{b} \rangle \right) \]

where \( \hat{\ell} \) is the look vector to a point and \( \hat{\ell}_o \) is the corresponding look vector to the flat surface at the same range.

- Making the usual 2 dimensional simplifications

\[ \phi_{flat} = \frac{4\pi}{\lambda} b \left( \sin(\theta - \alpha) - \sin(\theta_o - \alpha) \right) \]

\[ \approx \frac{4\pi}{\lambda} b \cos(\theta - \alpha) \delta \theta \]
From the figure we have

\[ \delta \theta \approx \frac{h_T}{\rho \sin \theta_o} \]

which gives

\[ \phi_{flat} = \frac{4\pi}{\lambda} b_{\perp} \delta \theta \]

\[ = \frac{4\pi}{\lambda} b_{\perp} \frac{h_T}{\rho \sin \theta_o} \]

Thus the topographic portion of the interferometric phase is a function of the perpendicular baseline length.
Sensitivity of Height with Respect to Phase

Sensitivity to Phase

\[
\frac{\partial \tilde{T}}{\partial \phi} = \frac{-\lambda \rho}{2\pi pb \cos(\theta - \alpha)} \begin{bmatrix} 0 \\ \cos(\theta) \\ \sin(\theta) \end{bmatrix}
\]

Ambiguity Height

\[h_a = 2\pi \frac{\partial \tilde{T}_z}{\partial \phi} = \frac{-\lambda \rho \sin(\theta)}{pb \cos(\theta - \alpha)}\]

\[p=1,2\]

- Observe that \(\frac{\partial \tilde{T}}{\partial \phi}\) is parallel to \(\hat{l} \times \hat{\nu}\).
3-D Height Reconstruction

• The full three dimensional height reconstruction is based on the observation that the target location is the intersection locus of three surfaces
  - range sphere \[ |\vec{P} - \vec{T}| = \rho \]
  - Doppler cone \[ f = \frac{2}{\lambda} \langle \vec{v}, \hat{\ell} \rangle \]
  - phase cone* \[ \phi = -\frac{2\pi p}{\lambda} \langle \vec{b}, \hat{\ell} \rangle \]

  Doppler and phase cones give two angles defining spherical coordinate system

• The cone angles are defined relative to the generating axes determined by
  - velocity vector          Doppler cone
  - baseline vector         phase cone

* Actually the phase surface is a hyperboloid, however for most applications where the phase equation above is valid, the hyperboloid degenerates to a cone.
Height Reconstruction Geometry

Scatterer is at intersection of Range Sphere, Doppler Cone and Phase Cone
Shuttle Radar Topography Mission
Mission Overview

Launch
Feb 11, 2000 - STS99

12 Tbytes data recorded on-board on 330 tape cassettes

Data returned with Shuttle to Ground Data Processing Facility

Three year processing

NIMA data validation, editing and distribution to military users

EDC for public distribution

Digital elevation data delivered in 1°x1° mosaiced cells

225 km C-band interferometric swaths map all landmass between ±60° latitude at least twice
SRTM Patch Processing Example

- Burst Interferogram Brightness Images
- Burst Interferogram Phase Images
- Uncalibrated Strips
- Patch Interferogram Brightness Images
- Patch Interferogram Phase Images
Motion compensation is required to account for boom dynamics as well as shuttle attitude changes. Left uncompensated these motions would generate hundreds of meters of height error.
SRTM Hardware Configuration

AODA

STA looks at stars

ATT looks at OTA

EDMs look at corner reflector array

GPS antenna (1 of 2)

flight segment

ground segment

global GPS ground receiver network (accessible via Internet)
SRTM Resolution Improvement

GTOPO30 DEM

SRTM DEM with radar image overlay

Lake Balbina, Brazil
SRTM Views Global Landscapes
SRTM Performance Summary

• Based on over hundreds of millions of comparisons, SRTM has an absolute height accuracy of 9.0m or better over a 1°x1° cell, at the 90% confidence level.

• Using the kinematic GPS data, probably the best ground truth data available, SRTM meets the absolute height requirement by a factor of 3.

• Both the absolute and relative SRTM height accuracy requirements are met.

• Using the kinematic GPS data, SRTM’s horizontal accuracy is better than 10 meters and we believe the results are consistent with no horizontal offset.

• Both the absolute and relative SRTM horizontal accuracy requirements are met.
Kinematic GPS

Fit Mean = 0.76m
Mean = 0.71m
Fit $\sigma$ = 3.09m
$\sigma$ = 3.35m
90% CL = 5.31m
Data Collection Options

For single pass interferometry (SPI) both antennas are located on the same platform. Two modes of data collection are common:

• *single-antenna-transmit mode* - one antenna transmits and both receive

• *ping-pong mode* - each antenna transmits and receives its own echoes effectively doubling the physical baseline.

\[
\phi = \frac{2\pi}{\lambda} (\rho_2 + \rho_1) - \frac{2\pi}{\lambda} (\rho_1 + \rho_1)
\]

\[
= \frac{2\pi}{\lambda} (\rho_2 - \rho_1)
\]

**Classic**

\[
\phi = \frac{2\pi}{\lambda} (\rho_2 + \rho_2) - \frac{2\pi}{\lambda} (\rho_1 + \rho_1)
\]

\[
= \frac{4\pi}{\lambda} (\rho_2 - \rho_1)
\]

**Ping-Pong**
Interferometric data can also be collected in the repeat pass mode (RPI). In this mode two spatially close radar observations of the same scene are made separated in time. The time interval may range from seconds to years. The two observations may be made different sensors provided they have nearly identical radar system parameters.

\[ \phi = \frac{2\pi}{\lambda} (\rho_2 + \rho_2) - \frac{2\pi}{\lambda} (\rho_1 + \rho_1) \]

\[ = \frac{4\pi}{\lambda} (\rho_2 - \rho_1) \]
Differential Interferometry

When two observations are made from the same location in space but at different times, the interferometric phase is proportional to any change in the range of a surface feature directly.

\[ \Delta \phi = \frac{4\pi}{\lambda} (\rho(t_1) - \rho(t_2)) = \frac{4\pi}{\lambda} \Delta \rho_{\text{change}} \]
Differential Interferometric Phase

\[ \phi = \frac{2\pi p}{\lambda} (\rho_2 - \rho_1) = \frac{2\pi p}{\lambda} \left( |\vec{l}_2| - |\vec{l}_1| \right) \]

\[ = \frac{2\pi p}{\lambda} \left( \langle \vec{l}_2, \vec{l}_2 \rangle^{\frac{1}{2}} - \rho_1 \right) \]

\[ \phi = \frac{4\pi}{\lambda} \left( \langle \vec{l}_1 + \vec{D} - \vec{b}, \vec{l}_1 + \vec{D} - \vec{b} \rangle^{\frac{1}{2}} - \rho_1 \right) \]

Assuming that

\[ |\vec{b}| \leq \rho \]

\[ |\vec{D}| \leq \rho \]

\[ |\langle \vec{b}, \vec{D} \rangle| \leq \rho \]

yields

\[ \phi = \frac{4\pi}{\lambda} \left( -\langle \vec{l}, \vec{b} \rangle + \langle \vec{l}, \vec{D} \rangle \right) \]
Differential Interferometry and Topography

• Generally two observations are made from different locations in space and at different times, so the interferometric phase is proportional to topography and topographic change.

\[ \phi = \frac{4\pi}{\lambda} \left( -\langle \hat{\ell}, \vec{b} \rangle + \langle \hat{\ell}, \vec{D} \rangle \right) \]

Topography Term
Change Term

If topography is known, then the second term can be eliminated to reveal surface change.

\[ \phi = \frac{4\pi}{\lambda} (\Delta \rho_{change} - \Delta \rho_{topo}) \]

\[ \phi = \frac{4\pi}{\lambda} (\Delta \rho_{change} - b \sin(\theta - \alpha)) \]

\[ \phi_{flat} = \frac{4\pi}{\lambda} \left( \Delta \rho_{change} - \frac{b \perp h_T}{\rho \sin \theta} \right) \]

Note: Sensitivity of phase with respect to change is much greater than with respect to topographic relief.
Extracting the Deformation Term - Pre-Existing DEM

\[ \phi_{b_1} = \frac{4\pi}{\lambda} \left( -\langle \hat{\ell}, \vec{b}_1 \rangle + \langle \hat{\ell}, \vec{D} \rangle \right) \]

Phase Simulated from DEM

\[ \phi_{topo} = -\frac{4\pi}{\lambda} \langle \hat{\ell}, \vec{b}_1 \rangle \]

Using Unflattened Phases

\[ \Delta \rho_{disp} = \langle \hat{\ell}, \vec{D} \rangle = \frac{\lambda}{4\pi} (\phi_{b_1} - \phi_{topo}) \]

Using Flattened Phases

\[ \Delta \phi_{flat_1} = \frac{4\pi}{\lambda} b_{\perp 1} \frac{h_T}{\rho \sin \theta_o} + \frac{4\pi}{\lambda} \Delta \rho_{disp} \]

\[ \Delta \phi_{flat_2} = \frac{4\pi}{\lambda} b_{\perp 1} \frac{h_T}{\rho \sin \theta_o} \]

\[ \Delta \rho_{disp} = \frac{\lambda}{4\pi} (\Delta \phi_{flat_1} - \Delta \phi_{flat_2}) \]
Extracting the Deformation - Three Passes

\[
\phi_{b_1} = \frac{4\pi}{\lambda} \left( -\langle \hat{\ell}, \vec{b}_1 \rangle + \langle \hat{\ell}, \vec{D} \rangle \right) \quad \text{Change Pair}
\]

\[
\phi_{b_2} = -\frac{4\pi}{\lambda} \langle \hat{\ell}, \vec{b}_2 \rangle \quad \text{Topo Pair}
\]

Using Unflattened Phases

\[
\Delta \rho_{disp} = \langle \hat{\ell}, \vec{D} \rangle
\]

\[
= \frac{\lambda}{4\pi} \left( \phi_{b_1} - \frac{\langle \hat{\ell}, \vec{b}_1 \rangle}{\langle \hat{\ell}, \vec{b}_2 \rangle} \phi_{b_2} \right)
\]

\[
= \frac{\lambda}{4\pi} \left( \phi_{b_1} - \frac{b_{\parallel_1}}{b_{\parallel_2}} \phi_{b_2} \right)
\]

Using Flattened Phases

\[
\Delta \phi_{flat_1} = \frac{4\pi}{\lambda} b_{\perp_1} \frac{h_T}{\rho \sin \theta_o} + \frac{4\pi}{\lambda} \Delta \rho_{disp}
\]

\[
\Delta \phi_{flat_2} = \frac{4\pi}{\lambda} b_{\perp_2} \frac{h_T}{\rho \sin \theta_o}
\]

\[
\Delta \rho_{disp} = \frac{\lambda}{4\pi} \left( \Delta \phi_{flat_1} - \frac{b_{\perp_1}}{b_{\perp_2}} \Delta \phi_{flat_2} \right)
\]
Differential Interferometry Sensitivities

• The reason differential interferometry can detect millimeter level surface deformation is that the differential phase is much more sensitive to displacements than to topography.

\[
\frac{\partial \phi}{\partial h} = \frac{2\pi p b \cos(\theta - \alpha)}{\lambda \rho \sin \theta} = \frac{2\pi p b_{\perp}}{\lambda \rho \sin \theta}
\]

Topographic Sensitivity

\[
\frac{\partial \phi}{\partial \Delta \rho} = \frac{4\pi}{\lambda}
\]

Displacement Sensitivity

\[
\sigma_{\phi_{\text{topo}}} = \frac{\partial \phi}{\partial h} \sigma_h = \frac{4\pi}{\lambda} \frac{b_{\perp}}{\rho \sin \theta} \sigma_h
\]

Topographic Sensitivity Term

\[
\sigma_{\phi_{\text{disp}}} = \frac{\partial \phi}{\partial \Delta \rho} \sigma_{\Delta \rho} = \frac{4\pi}{\lambda} \sigma_{\Delta \rho}
\]

Displacement Sensitivity Term

Since \( \frac{b}{\rho} \ll 1 \) \( \Rightarrow \) \( \frac{\sigma_{\phi_{\text{disp}}}}{\sigma_{\Delta \rho}} \gg \frac{\sigma_{\phi_{\text{topo}}}}{\sigma_h} \)
Strengths and Limitations

• The major strengths for using radar differential interferometry are
  – Wide area coverage
  – Millimeter scale accuracy locally
  – Geophysically useful even without other data sets
  – Complementary to established geodetic and seismic tools (e.g. GPS and leveling)

• Although there has been great success using radar differential interferometry for deformation measurement its application can be limited by
  – Temporal decorrelation of the surface
  – Other surface changes
  – Atmospheric and ionospheric effects
  – Poor DEM availability and Quality or lack of DEM pair
Some Examples of Deformation

Hector Mine Earthquake

Etna Volcano
Example of SRTM DEM Improvement

Differential Interferometry

Mount Etna inflation signature cleaned up considerably by using SRTM data to remove topographic signature.
West Antarctic Ice Streams from InSAR

The time evolution of ice stream flow variability is uniquely imaged by InSAR. Complete coverage by InSAR is needed to understand flow dynamics of the potentially unstable marine ice sheet.

Joughin et al., 1999
Along Track Interferometry (ATI)

• By having antennas separated in the along track direction interferometry provides a very sensitive measure of the line-of-sight velocity.

• Ocean currents and ship velocities can be measured using along track interferometry.

$$\phi = \frac{4\pi}{\lambda} \Delta \rho = \frac{4\pi}{\lambda} \frac{\partial \rho}{\partial t} \Delta t = \frac{4\pi}{\lambda} V_{los} \frac{D}{V_{spc}}$$
ATI Experiment: Straits of Juan de Fuca

C-Band (AF/AA)

L-Band (AF/AA)
Phase Measurements in Interferometry

\[ \phi_{\text{top}} = \frac{4\pi}{\lambda} (\rho_1 - \rho_2) \]

\[ \phi_{\text{top}} \approx \frac{4\pi}{\lambda} \overline{B} \cdot \overline{L} \]

\[ \phi_m = \text{mod} (\phi_{\text{top}}, 2\pi) \]

\[ \phi_{\text{unw}} = \phi_{\text{top}} + \phi_c \]
Radar Coherent Backscatter

Pixels in a radar image are a complex phasor representation of the coherent backscatter from the resolution element on the ground and the propagation phase delay.

\[ A_b e^{i\phi_b} \sum_i A_{ei} e^{j\phi_{ei}} e^{-j\frac{4\pi}{\lambda} \rho_{ei}} \]

Backscatter phase delay is coherent sum of contributions from all elemental scatterers in the resolution element with backscatter and their differential path delays.

\[ \rho_{ei} \]

Propagation Phase Delay \( \phi = -j \frac{4\pi}{\lambda} \rho \)
Interferometric Phase Characteristics

Pixels in two radar images observed from nearby vantage points have nearly the same complex phasor representation of the coherent backscatter from a resolution element on the ground but a different propagation phase delay.

\[ s_1 = A_b e^{j \phi_b} e^{-j \frac{4\pi}{\lambda} \rho_1} \]
\[ s_2 = A_b e^{j \phi_b} e^{-j \frac{4\pi}{\lambda} \rho_2} \]
Correlation Theory

Signals decorrelate due to

- Thermal and Processor Noise
- Differential Geometric and Volumetric Scattering
- Rotation of Viewing Geometry
- Random Motions Over Time

Decorrelation observed as phase standard deviation

- Affects height and displacement accuracy and ability to unwrap phase
- Can define an effective SNR as thermal SNR degraded by other decorrelation effects
Correlation Definition

For signals \( s_1 \) and \( s_2 \) observed at interferometer apertures 1 and 2, the correlation \( \gamma \) is given by

\[
\gamma = \frac{|\langle s_1 s_2^* \rangle|}{\sqrt{\langle s_1 s_1^* \rangle \langle s_2 s_2^* \rangle}}
\]

Deterministic signals or combinations have perfect correlation

\[
\gamma = \frac{| \langle |s_1|^2 e^{i\phi_1} |s_2|^2 e^{-i\phi_2} \rangle |}{\sqrt{|s_1|^2 |s_2|^2}} = 1
\]

Signals with random components have imperfect correlation

\[
\gamma = \frac{| \langle |s_{1i}|^2 e^{i\phi_{1i}} |s_{2i}|^2 e^{-i\phi_{2i}} \rangle_i |}{\sqrt{\langle |s_{1i}|^2 \rangle_i \langle |s_{2i}|^2 \rangle_i}} \neq 1
\]
Expectation Estimator

Expectation over the ensemble of realizations \( S_{1i} \) and \( S_{2i} \) cannot be calculated from a specific realization, that is, the observations \( S_1 \) and \( S_2 \). In general, an estimator derived from the image data must be devised.

The maximum likelihood estimator (MLE) of the interferometric phase difference between images forming an interferogram that have homogeneous backscatter and constant phase difference is

\[
\hat{\phi} = \arg \sum_{l,m}^{N,M} s_{1l,m} s_{2l,m}^*
\]

where \( l \) and \( m \) are image indices relative to some reference location and the sum is computed over a \( N \times M \) box. \( N \) and \( M \) are known as the number of looks in their respective image dimensions.
Correlation Estimator

The standard estimator of the interferometric correlation between images forming an interferogram that have homogeneous backscatter and constant phase difference is

\[ \gamma = \frac{\sum_{l,m} S_{1l,m} S_{2l,m}^*}{\sqrt{\sum_{l,m} S_{1l,m}^2 \sum_{l,m} S_{2l,m}^2}} \]

This estimator is biased. As an example, consider \( N = 1, M = 1 \). Then

\[ \gamma = \frac{S_{1x,y} S_{2x,y}^*}{\sqrt{S_{1x,y}^2 S_{1x,y}^* S_{2x,y}^* S_{2x,y}^*}} = 1 \]

where \( x \) and \( y \) are the coordinates of a particular image pixel. Other biases arise when backscatter homogeneity and phase constancy are violated.
Correlation Estimator Characteristics

A general set of curves reveals the nature of the estimator bias when backscatter is homogeneous and phase is constant.
Relationship of Phase Noise and Decorrelation

- Increased decorrelation is associated with an increase in the interferometric phase noise variance.
- If averaging $N > 4$ samples, the phase standard deviation approaches the Cramer-Rao bound on the phase estimator:

\[
\sigma_\phi = \frac{1}{\sqrt{2N}} \frac{\sqrt{1 - \gamma^2}}{\gamma}
\]

Phase noise contributes to height errors in interferometry as demonstrated in the sensitivity equations.
Thermal Noise Decorrelation

Radar receiver electronics will add thermally generated noise to the image observations.

\[ s_1' = A_b e^{j\phi_b} e^{-j\frac{4\pi}{\lambda} \rho_1} + n_1 \; ; \; s_2' = A_b e^{j\phi_b} e^{-j\frac{4\pi}{\lambda} \rho_2} + n_2 \]

The added noise contributes randomly to the interferometric phase from pixel to pixel, causing thermal noise decorrelation. Assuming uncorrelated noise, the correlation between \( s_1 \) and \( s_2 \) is

\[
\gamma = \frac{\left| \langle (s_1 + n_1)(s_2 + n_2)^* \rangle \right|}{\sqrt{\langle (s_1 + n_1)(s_1 + n_1)^* \rangle \langle (s_2 + n_2)(s_2 + n_2)^* \rangle}}
\]

\[
= \frac{|s_1 s_2|}{\sqrt{\langle s_1^2 + n_1^2 \rangle \langle s_2^2 + n_2^2 \rangle}}
\]

\[
= \frac{\sqrt{P_1} \sqrt{P_2}}{\sqrt{P_1 + N_1} \sqrt{P_2 + N_2}} = \frac{1}{\sqrt{1 + N_1 / P_1}} \frac{1}{\sqrt{1 + N_2 / P_2}}
\]
Thermal Noise Decorrelation

The correlation is related in a simple way to the reciprocal of the Signal-to-Noise Ratio (SNR). For observations with identical backscatter and equal noise power,

\[ \gamma = \frac{1}{1 + \frac{N}{P}} = \frac{1}{1 + \text{SNR}^{-1}} \]

Decorrelation is defined as

\[ \delta = 1 - \gamma \]

The decorrelation due to thermal noise can vary greatly in a scene, not from thermal noise variations, but from variations in backscatter brightness. Extreme cases are:

• Radar shadow, where no signal is returns; correlation is zero.
• Bright specular target, where signal dominates return; correlation is 1
SNR from the Radar Equation

Commandable Radar Parameters

- $\tau$ = pulse width (affects swath width and SNR)
- $\rho_{\text{start}}$ = start of data collection window (affects starting look angle)
- DWP = length of data collection window (affects swath width)
- $B$ = chirp bandwidth (affects range resolution)

$$SNR = \frac{P_t G_t G_r}{L_t L_r kT_{eff} B_n} \frac{\lambda^2}{(4\pi)^3} \rho^4 \left( \frac{c\tau}{2 \sin(\theta_i)} \right) \left( \rho \theta_{az}^{eff} \right) \sigma_o$$

$P_{sys}$ - lumped effect of transmit gain and various losses

$N = kT_{eff} B_n$

where

$T_{eff} = \text{Effective Receiver Temperature}$

$B_n = \text{System Noise Bandwidth}$
SNR from the Radar Equation

Commandable Radar Parameters
- $\tau$ = pulse width (affects swath width and SNR)
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$$SNR = \frac{P_t G_t G_r}{L_t L_r kT' B_n} \left( \frac{\lambda^2}{4\pi} \right)^3 \rho^4 \left( \frac{c\tau}{2 \sin(\theta_i)} \right) \left( \rho \theta_{\text{az}}^{\text{eff}} \right) \sigma_o$$

$P_{\text{sys}}$ - lumped effect of transmit gain and various losses

$$N = kT' B_n$$
where
$T'$ = Effective Receiver Temperature
$B_n$ = System Noise Bandwidth
Baseline Decorrelation

Pixels in two radar images observed from nearby vantage points have nearly the same complex phasor representation of the coherent backscatter from a resolution element on the ground.

As interferometric baseline increases, the coherent backscatter phase becomes increasingly different randomly, leading to “baseline” or “speckle” decorrelation.

\[ s_1 = A_{b1} e^{j\phi_{b1}} e^{-\frac{j4\pi}{\lambda} \rho_1} \]
\[ s_2 = A_{b2} e^{j\phi_{b2}} e^{-\frac{j4\pi}{\lambda} \rho_2} \]
Baseline Decorrelation: Resolution Cell

Interferometric geometry for a horizontal baseline in the cross-track plane

\[ \rho \]

\[ \theta_1 \]

\[ \theta_2 \]

\[ y \sin \theta_1 \]

\[ y \sin \theta_2 \]

Origin of resolution cell
Baseline Correlation Formulation

Consider signals from the two antennas displaced in angle:

\[ s_1(x_0, y_0) = \int dx \int dy f(x - x_0, y - y_0) e^{-j \frac{4\pi}{\lambda}(\rho + y \sin \theta_1)} W(x, y) + n_1(x_0, y_0) \]

\[ s_2(x_0, y_0) = \int dx \int dy f(x - x_0, y - y_0) e^{-j \frac{4\pi}{\lambda}(\rho + y \sin \theta_2)} W(x, y) + n_2(x_0, y_0) \]

Cross-correlation yields:

\[ s_1 s_2^* = \int dx \int dy \int dx' \int dy' f(x - x_0, y - y_0) f^*(x' - x_0, y' - y_0) \]

\[ e^{-j \frac{4\pi}{\lambda}(y \sin \theta_1 - y' \sin \theta_2)} W(x, y) W^*(x', y') \]

plus noise cross-products

\[ f(x,y) = \text{the surface reflectivity function} \]

\[ W(x,y) = \text{the imaging point spread function} \]
Surface Assumption for Correlation Formulation

If the surface is modeled as *uniformly distributed, uncorrelated* scattering centers, that is:

\[
\langle f(x, y) f^*(x', y') \rangle = \sigma_0 \delta(x - x', y - y')
\]

where \( \sigma_0 \) is the mean backscatter cross section per unit area from the surface.

Under this assumption, the correlation expression reduces to:

\[
\langle s_1 s_2^* \rangle = \sigma_0 \int dx \int dy |W(x, y)|^2 e^{-j \frac{4\pi}{\lambda} y \cos \theta} \Delta \theta
\]

The shape of the correlation function as a function of angle is the Fourier Transform of the Image Point Spread Function.
Form of Baseline Correlation Function

If $W(x,y)$ is assumed to be a sinc function, then the integral can be done in closed form (for ping-pong mode)

$$\gamma_B = 1 - \frac{2(B \cos \theta)(\Delta \rho_y \cos \theta)}{\lambda \rho}$$

$$= 1 - \frac{2B_\perp \Delta \rho_\perp}{\lambda \rho}$$

For $B >> \rho$

$$\Delta \theta \sim \frac{B_\perp}{\rho}$$

This function goes to zero at the critical baseline

$$B_{\perp,\text{crit}} = \frac{\lambda \rho}{n \Delta \rho_\perp}$$

$n = 1, 2$
The “Pixel Antenna” View of Baseline Decorrelation

Each resolution element can be considered a radiating antenna with beamwidth of $\Delta \theta_{\Delta \rho_{\perp}}$, which depends on the range and local angle of incidence.

$$\Delta \theta = \frac{\lambda}{2\Delta \rho_{\perp}}$$

When the two apertures of the interferometer are within this beamwidth, coherence is maintained. Beyond this critical baseline separation, there is no coherence for distributed targets.
Overcoming Baseline Decorrelation

Distributed targets correlate over a narrow range of baselines

Effective Projected Resolution

Pixels dominated by single scatterers generally behave though imaged at much finer resolution
Critical Baseline

The critical baseline is the aperture separation perpendicular to the look direction at which the interferometric correlation becomes zero.

\[
B_{\text{crit}} = \rho \Delta \theta \Delta \rho_\perp = \frac{\lambda \rho \tan \theta}{n \Delta \rho} \quad n=1,2
\]

Interferometers with longer wavelengths and finer resolution are less sensitive to baseline decorrelation. When the critical baseline is reached, the interferometric phase varies as

\[
\frac{\partial \phi_{\text{crit}}}{\partial \rho} = \frac{\partial \phi_{\text{crit}}}{\partial \theta} \frac{\partial \theta}{\partial \rho}
\]

\[
= \frac{2n\pi}{\lambda} B_{\text{crit}} \frac{1}{\rho \tan \theta} = \frac{2\pi}{\Delta \rho}
\]

The relative phase of scatterers across a resolution element changes by a full cycle, leading to destructive coherent summation.
Rotational Decorrelation

Rotation of scatterers in a resolution element can be thought of as observing from a slightly different azimuthal vantage point. As with baseline decorrelation, the change in differential path delay from individual scatterers to the reference plane produces rotational decorrelation.

\[ s_1 = A_b e^{j\phi_b} e^{-j\frac{4\pi}{\lambda} \rho_1} \]
\[ s_2 = A_b e^{j\phi_b} e^{-j\frac{4\pi}{\lambda} \rho_2} \]

The critical rotational baseline is the extent of the synthetic aperture used to achieve the along track resolution.
Rotational Correlation Formulation

Cross-correlation for this case is:

\[ s_1 s_2^* = \int dx \int dy \int dx' \int dy' f(x - x_0, y - y_0) f^*(x' - x_0, y' - y_0) \]
\[ \times e^{-i \frac{4\pi}{\lambda} \delta \sin \theta (\sin \theta_1 - \sin \theta_2)} W(x, y) W^*(x', y') \]

plus noise cross-products
Form of Rotational Correlation Function

A similar Fourier Transform relation as found in the baseline decorrelation formulation exists

\[ \gamma_\phi = 1 - \frac{n \sin \theta B_\phi R_x}{\lambda \rho} \]

\[ n = 1, 2 \]

where \( R_x \) is the azimuth resolution and \( B_\phi \) is the distance along track corresponding to the rotation angle of the look vector.

This function goes to zero at the critical rotational baseline

\[ B_{\phi, \text{crit}} = \frac{\lambda \rho}{n \Delta \rho_\phi}, \Delta \rho_\phi \equiv R_x \sin \theta \]
Scatterer Motion

Motion of scatterers within the resolution cell from one observation to the next will lead to randomly different coherent backscatter phase from one image to another, i.e. “temporal” decorrelation.

\[ A_b(t_1)e^{j\phi_b(t_1)} \]

\[ A_b(t_2)e^{j\phi_b(t_2)} \]

Initial Observation

Image Pixel/Resolution Element

Some time later

Propagation Phase Delay \[ \phi = -j\frac{4\pi}{\lambda}\rho \]
Motion Correlation Formulation

Assume the scatterers move in the cross-track and vertical

\[ s_1 s_2^* = \int dx \int dy \int dz \int dx' \int dy' \int dz' f(x-x_0, y-y_0, z-z_0) f^*(x'-x_0, y'-y_0, z'-z_0) \]

\[ e^{-j \frac{4\pi}{\lambda} (\delta y \sin \theta + \delta z \cos \theta)} W(x, y, z) W^*(x', y', z') \]

plus noise cross-products

If the motions are independent and unrelated to position

\[ \langle s_1 s_2^* \rangle = \sigma_0 \int d\delta x \int d\delta y \ e^{-j \frac{4\pi}{\lambda} (\delta y \sin \theta + \delta z \cos \theta)} p_y(\delta y)p_z(\delta z) \]

where \( p_y \) and \( p_z \) are the probability distributions of the scatterer locations
Form of Motion Correlation Function

The Fourier Transform relation can be evaluated if Gaussian probability distributions for the motions are assumed

\[ \gamma_t = e^{-\frac{1}{2} \left( \frac{4\pi}{\lambda} \right)^2 \left( \sigma_y^2 \sin^2 \theta + \sigma_z^2 \cos^2 \theta \right)} \]

where \( \sigma_{y,z} \) is the standard deviation of the scatterer displacements cross-track and vertically

Note correlation goes to 50% at about 1/4 wavelength displacements
Repeat-pass Interferometry and Coherent Change Detection
L-Band low frequency improves correlation in vegetated areas

Most radars do well in areas of sparse vegetation

But maintaining correlation in dense vegetation requires longer wavelengths

Loss of correlation is due to:
- volume of vegetation
- movement of vegetation
- dielectric change (moisture)

Effective phase center

VHF L-band
UHF C-band
P-band X-band
Coherent Change Detection
SIR-C L and C-band Interferometry

6 month time separated observations to form interferograms
Simultaneous C and L band

InSAR experiments have shown good correlation at L-band
Airborne InSAR experiments have shown good correlation at L-band
A Correlation Test:
What were the interferometric observation conditions?
Correlation in Practical Systems

• For Single-Pass Two-Aperture Interferometer Systems
  – System noise and baseline/volumetric decorrelation are particularly relevant
  – Proper design of system can control
    • noise, by increasing SNR and minimizing shadow
    • baseline decorrelation, by choosing proper wavelength and geometry
  – Volumetric decorrelation can be inferred and used as a diagnostic of the vertical distribution of scatterers

• For Multi-Pass Single-Aperture Interferometer Systems
  – All decorrelation effects may occur
  – Time series analysis and some assumptions may be useful in distinguishing decorrelation contributors
For a calibrated interferometer, $\gamma_{\text{SNR}}$ can be calculated for each pixel. For single-pass systems, $\gamma_t = 1$ and $\gamma_\phi$ is usually nearly unity. The baseline decorrelation can be computed from the known geometry, leaving the volumetric correlation, which can be used as a diagnostic of surface scatterers.

For calibrated repeat-pass systems with no rotational decorrelation, the product $\gamma_V \gamma_t$ is separable only if $\gamma_t \neq 0$ and $\gamma_V$ estimates from several baselines are compared.
Interferometric Error Sources

- Interferometric Decorrelation

In addition to the decorrelation contributions, several other sources of error exist in interferometry. These include:

- Layover and shadow in radar imagery from slant range geometry
- Multiple scattering within and among resolution cells
- Range and Azimuth sidelobes due to bandwidth/resolution constraints
- Range and azimuth ambiguities due to design constraints
- Multipath and channel cross-talk noise as low-level interference
- Phase unwrapping errors
- Calibration errors
- Propagation delay errors from atmosphere and ionosphere
In addition to baseline and position, time and phase delays in the radar require calibration.
Calibration Needs for an Interferometer

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Element</th>
<th>Accuracy</th>
</tr>
</thead>
<tbody>
<tr>
<td>Baseline vector, $B$, including length and attitude, for reduction of interferometric phase to height</td>
<td>Locations of the phase centers of both antennas</td>
<td>a few millimeters</td>
</tr>
<tr>
<td>Absolute radar range from one antenna to targets, for geolocation</td>
<td>Time delay through the composite transmitter/receiver</td>
<td>a few nanoseconds</td>
</tr>
<tr>
<td>Differential radar range between channels, for image alignment in interferogram formation</td>
<td>Time delays through the receiver chains (but not the transmitter chain)</td>
<td>fewer nanoseconds</td>
</tr>
<tr>
<td>Differential phase between channels, for determination of the topography</td>
<td>Phase delays through the receiver chains (but not the transmitter chain)</td>
<td>a few degrees</td>
</tr>
</tbody>
</table>
Hawaii SIR-C One Day Repeat Pass Data Showing Tropospheric Distortion
Interferometric Processing Flow

SAR Signal Processing

Module for which exact position in processing is algorithm dependent

Ancillary Data

Used if needed by selected algorithm

Raw Signal Data Antennna 1

Decode and Range Compression

Azimuth Compression

Image Resample and Interferogram Formation

Looks

Raw Signal Data Antennna 2

Decode and Range Compression

Azimuth Compression

Looks

Absolute Phase Determination

Phenomenological Studies

Baseline Measurement

Troposphere Model

Post Processing

Coarse DTE

Primary Output Product Generation

Mosaicking

3 Dimensional Location and Gridding

NAV Data

Phase Unwrapping

SAR Signal Processing
Repeat Pass Interferometry Processing

Procedure to pre-determine offsets: process first and last piece of data, then interpolate offsets between.
Scene Matching and Coregistration

Cross-correlate images by expedient method

- Tie-points by eye
- Fringe visibility
- Amplitude correlation
- Amplitude difference

Fit a 2-D function (polynomial) to $(\Delta \rho, \Delta a)$ offset field
Scene Matching on Imagery

- There are a plethora of algorithms in existence for automatic scene matching, each with particular strengths and weaknesses.
- SAR scenic matching for interferometry applications is difficult for several reasons:
  - Thermal noise causes the fine structure of two SAR images to differ, adding noise to the correlation measurements.
  - Geometric speckle noise is similar in images acquired in nearly the same geometry, but as the interferometric baseline increases, differing speckle noise corrupts the matching correlation.
  - Scene decorrelation is another form of speckle noise difference that corrupts matching correlation.
- SAR scenic matching for mosaicking applications involves greater challenges, including severe speckle noise differences, layover and shadow effects.
Automatic Scene Matching

- Find overlap region and sample points at specified spacing in along track and cross track direction.
- Typical window sizes are 64x64 pixels for image data and 64x64 or 128x128 pixels for height data.
- Uses a modified Frankot's method for rejecting bad matches and to provide an estimate of the match covariance matrix.
- Cross correlation uses a normalized mean correlation function, mean of search window is calculated over the region in common with the reference window.

Mean of reference is constant

Reference Window

Mean of search varies with position

Search Window
Match Correlation Estimate

- Correlation is computed in the spatial domain using a normalized cross correlation algorithm.
- Let $I_1(\bar{x})$ be the image values at a point $\bar{x}$ in the reference window in the first data set, $\bar{I}_1$, the mean of the intensities in that window.
- Let $I_2(\bar{x} + \bar{d})$ be the image value at point $\bar{x} + \bar{d}$ in the search window of the second data set, and $\bar{I}_2(\bar{d})$ the mean of the intensities in that window.
- Viewing each image as a vector in an n-dimensional vector space then the cross correlation is computed as

$$c(\bar{d}) = \frac{\langle I_1 - \bar{I}_1, I_2(\bar{d}) - \bar{I}_2(\bar{d}) \rangle}{\sqrt{\|I_1 - \bar{I}_1\| \|I_2(\bar{d}) - \bar{I}_2(\bar{d})\|}} = \frac{\langle I_1 - \bar{I}_1, I_2(\bar{d}) - \bar{I}_2(\bar{d}) \rangle}{\sigma_{I_1} \sigma_{I_2(\bar{d})}}$$

where $\sigma_{I_1}$ and $\sigma_{I_2(\bar{d})}$ are the standard deviations of the image intensities in data sets 1 and 2, respectively.

Space-domain correlation used for speed and because large irregular data gaps are trouble free.
Baseline Estimation from Offsets

- Baseline estimation in repeat pass interferometry is a rich subject. Briefly:
- Offset field determined from scene matching carries information sufficient to reconstruct the baseline to an accuracy commensurate with ability to co-register images.
- For nearly parallel, smoothly varying, orbit tracks, the baseline can be modeled as a simple function of the range $\rho$

$$B_\parallel(\rho) \approx B \sin(\theta_0(\rho) - \alpha) = \Delta \rho(\rho)$$

$$B_\perp(\rho) \approx B \cos(\theta_0(\rho) - \alpha) = \frac{d \Delta \rho}{d \theta}(\rho)$$

- Solve two equations for two unknowns: $B, \alpha$
- Evaluating baseline at several widely-spaced along-track (azimuth) locations gives azimuth history of the baseline.
Converging Baselines

• For orbit tracks with substantial convergence, azimuth offsets strongly characterize convergence rates.

**Observation vector:**

\[ \{ \cdots, \Delta \rho(\rho_i, a_i), \Delta a(\rho_i, a_i), \Delta \rho(\rho_{i+1}, a_{i+1}), \Delta a(\rho_{i+1}, a_{i+1}), \cdots \} \]

**Parameter vector:**

\[ \{ B_{s0}, B_{c0}, B_{h0}, \gamma, \frac{\partial B_c}{\partial s}, \frac{\partial B_h}{\partial s} \} \]

where \( \gamma \) is an image scaling factor resulting from velocity differences between the tracks, and \((s,c,h)\) are local coordinates defined by the orbit track (see notes on coordinate systems).
Baseline Accuracy from Offsets

- Baselines determined from offset fields are typically accurate to a fraction of the scene matching accuracy that depends on the model function and the number of offset estimates. Example:
  - Zero baseline image pair with 10 match points across range.
  - Scene matching accuracy of 1/20 pixel resolution at typical single look resolution of 15 meters: $75 \text{ cm} / 3 = 25 \text{ cm}$

- This accuracy is insufficient for topographic mapping

\[
\sigma_h = \frac{\rho}{B} \sin \theta \cos \alpha \sigma_{B_z}
\]

\[
= 320 \text{ m} \ ; \ \rho = 830 \text{ km}, \ B = 250 \text{ m}, \ \theta = 23^\circ, \ \alpha = 0, \ \sigma_{B_z} = 25 \text{ cm}
\]

These geometric parameters are typical of ERS with baselines suitable for topography. Generally, ground control points tied to the unwrapped interferometric phase are required for mapping.
Interferogram Formation

From interferogram and detected imagery, correlation can be formed properly at any resolution: *i.e. average of correlation is not correlation of average*
Registration Implementation

• In resampling single look complex image to register properly with the reference image, care must be taken in interpolation of complex data
  – The azimuth spectrum of squinted SAR data is centered at the Doppler centroid frequency - a band-pass signal
  – Simple interpolators, such as linear or quadratic interpolators, are low-pass filters and can destroy band-pass data characteristics
• Band-pass interpolators or spectral methods preserve phase fidelity

![Signal Spectrum Diagram](image)
Repeat Pass Interferometry
Back-end Processing

Interferogram

Detected Imagery

Remove Reference Fringes

Unwrap Phase

Replace Reference Fringes

Correlation File and Mask

Geolocation

Estimate Baseline

Single-look Correlation Offset Estimates

DEM/Topo

Performance Summary:
L-band: good over most terrain
C-band: poor over most terrain
N-pass Differential Processing

Short Baseline Interferogram
- Detected Imagery
  - Correlation File and Mask
  - Remove Reference Fringes
  - Unwrap Phase

Long Baseline Interferogram
- Detected Imagery
  - Correlation File and Mask
  - Remove Reference Fringes
  - Unwrap Phase

\[ B_{1s} \]
\[ B_{ll} \]

Displacement
Amplitude
Correlations
Geolocation
2-Pass Differential Processing

Interferogram

Detected Imagery

Remove Reference Fringes

Unwrap Phase

Correlation File and Mask

Baseline Estimate

Simulate Interferogram Phase

DEM/Topo

Unwrap Phase

Geolocation

Displacements

Amplitude

Correlation

Range/Doppler -> Lat/Lon Mapping
Phase Unwrapping

- Elements of the phase unwrapping problem: From the measured, wrapped phase, unwrap the phase from some arbitrary starting location, then determine the proper ambiguity cycle.

\[
\phi = -\frac{4\pi}{\lambda}(\rho_1 - \rho_2) \equiv -\frac{4\pi}{\lambda} \cdot \hat{B} \cdot \hat{n}_\text{los}
\]

\[
\phi_{\text{meas}} = \text{mod}(\phi, 2\pi)
\]

\[
\phi_{\text{unw}} = \bigcup(\phi_{\text{meas}}) = \phi + 2\pi k
\]

where \(k\) is an integer to return the absolute phase
Two-Dimensional Phase Unwrapping

- Two dimensional phase field values below are in units of cycles.
- One-dimensional unwrapping criterion of half-cycle proximity is inconsistent in two dimensions.

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- Residues, marked with + and -, define ambiguous boundaries.
Residues in Phase Unwrapping

- The wrapping operator delivers the true phase modulo $2\pi$, in the interval $-\pi < \phi < \pi$.
- The true phase gradient is conservative: $\nabla \times \nabla \phi = 0$
- The wrapped gradient of the measured, wrapped phase, however, may not be conservative: $\nabla \times W\{\nabla \phi_w\} \neq 0$
- When this function is non-conservative, its integration becomes path dependent.
- Residues occur at locations of high phase noise and/or phase shear such that the wrapped gradient of the measured, wrapped phase is no longer conservative.

\[
\nabla \times \nabla \phi = 0
\]
\[
\nabla \times W\{\nabla \phi_w\} \neq 0
\]
Branch Cuts in Phase Unwrapping

- Branch-cut algorithms (Goldstein, Zebker, and Werner 1986) seek to neutralize these regions of inconsistency by connecting residues of opposite solenoidal sense with cuts, across which integration may not take place.
- Branch cut connections force path independence in the integration of the wrapped phase gradient.
- If done properly, the integrated phase field will be correct. But which is

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-0.4 -0.1 -0.1  0.2
-0.4 -0.1 -0.1  0.2
-0.4 -0.1 -0.1  0.2
```
Illustration of Branch Cut Algorithm

Connect all simple pairs

Expand search window and connect where possible

Initiate search from every residue encountered

Could not connect last residue with this search window size

Repeat all steps with larger search window size
Branch Cut Strategies

• The standard GZW algorithm is designed to connect residues into a neutral network into the shortest possible connection tree, i.e. to minimize the length of the individual branch cuts comprising a tree.

• This will not necessarily create the shortest possible tree, since GZW makes many unnecessary connections in its search for neutrality.

• Various criteria have been devised to place guiding centers (unsigned residues) along expected paths to facilitate the right choice in branch cut connection.
Guiding Center Criteria

• A number of criteria have been devised for selecting guiding centers, each more or less tailored to characteristics of SAR data:
  – when phase slope exceeds threshold (implies layover)
  – when derivative of phase slope exceeds threshold
  – when radar brightness exceeds threshold (implies layover)
  – when decorrelation estimator exceeds threshold (implies noise and/or layover)

• Some guiding center selections help in some cases
• Difficult to assess performance in a quantitative way
Ground control reference points can be used to determine the absolute phase ambiguity.
What’s Next

• Having been exposed to the principles and theory of InSAR, you will spend a day learning to use the roipac software and becoming familiar with
  – images, image pairs and registration offsets
  – interferograms and phase
  – correlation and noise
  – phase ramps and estimating baselines
  – deformation signals and error signals